



24 SRT

24 87 - plated

24 HT.

Ingraves -  
Bulkheads -

.016 } Thickness of skin  
.032 }  
.051 }

Copper - magnesium cover  
with aluminum - clad  
Steel taper fit - cadmium  
plated or stainless steel  
Housing of ~~air~~ <sup>medium</sup> ~~iron~~  
construction.

Station 3, first bulkhead,  
forward compartment. - 1-3

Cockpit. - 3-4

Bomb bay. 4-5

Radio comp.

Waste comp.

Hot hammer comp.

Ailerons have 1 actuating push pull rod.

Outer panel .032 skin, truss type ribs with corrugated skin for strength. .016 skin for trailing edge. 4 terminals with 10 taper pins and bolts in abutting edge of angle. accessories splice angle plates. outer panel 500<sup>th</sup> leading edge held on by pins with keeper and screws.

Mounts interchangeable made of steel, mounted by four bolts. never weld engine mounts. accessory compartment closed by walls of steel, no gap clearance required.  $\frac{1}{16}$  in. wall thickness

quadrant operating mechanism (11)

8 taper pins in 4 terminals hold vertical fins. Tail 19 ft high. order of assembling the sponage (vertical fin, rudder, rudder trim tab, horizontal stabilizer elevators and elevator trim tab).

400<sup>th</sup> torque on rudder mechanism.

Secure all cables before disconnecting cables to rudder and elevators.

Trim tabs must be down to adjust tensions of cables. (color code forward wall of accessories section.

bulkhead 8, ensuring gaps.)

automatic pilot has 3 servo units on elevator, rudder, ailerons.

90<sup>th</sup> tensions on servo units.

6 wires to inch broken, replace cable.  
to replace cable loosen turnbuckle  
remove spacers, take out pulley  
wheel, attach feed line.

### Remaining elevators

1. Loosen boot and fin fairing  
around torque tube.
2. Streamline elevator with a  
wooden clamp.
3. Lock control column  $10\frac{1}{2}^{\circ}$   
in level flight.
4. Release tensions on main  
elevator cables, 2 turnbuckles  
at bulkhead 6+2 at bulkhead  
no. 9.
5. Put tab in down position.

b. Clamp elevator tab cables behind fairleads, ahead of turnbuckle in waste comp.

7. Tape drums + elevators.

8. Tie lead line to cables for reassembly.

12 gear boxes on flap system

12 flap rods,  $45^\circ$  down position motor located in left wing between 1 and 2 rods.

With rods closed turn 48 turns out, then adjust stops 45 turns =  $45^\circ$  angle. always free gear boxes by disconnecting gear box and turn gears  $\frac{1}{2}$  to  $\frac{3}{4}$  turn in desired direction

Bus cables - 2 into 1 - with  
short and long turnbuckle  
-----  
Compass

Variation - angle of difference.

agonic line - no variation

Isogonic lines - equal magnetic  
variation at all points

Deviation - difference between  
compass heading and heading  
of the ship.

Navigator's compass is master  
compass. all good compasses  
are really master compasses.  
100 yds is fairly safe distance from  
magnetic disturbances.

Rubber line - stationing mark

card or spider: part that always  
points north, (azimuth is always  
marked of in degrees, rotates.)

(Compensating Compass. Swing  
compass N, E, S, W. taking out  
all error in 1st two directions,  
 $\frac{1}{2}$  error in last two directions.)

{ 5° tolerance in Navigator's compass  
{ 10° tolerance in Pilot's compass.

R.P.M. should be at least 1250  
to 1450 so that generator will  
cut in before compensating.

(Type D-12 compensating magnets)

(acid free water white kerosene  
Compass fluid)

## (Pitot Static Instruments.) -

air speed indicator - Impact and static.  
rate of climb - static pressure  
altimeter - .

Impact pressure - static pressure

(Setting altimeter for cross-country)

1. Learn field elevation.
2. Learn Barometric pressure reduced to sea level for that field at that time.
3. Set Barometric scale on the altimeter to agree with the tower reading.
4. Then hands should agree with field elevation  $\pm$  50 ft.

# Electricity

## Electrical Symbols

1. (P) D.C. Power circuits.
2. (I) Ignition
3. (PF) Propeller Feathering
4. (S) Starter
5. (A) Autogen instruments
6. (M) instruments.
7. (R) Retraction motors
8. (W) Warning signals
9. (D) De-ice and oil dilution
10. (IB) instrument lighting
11. (IL) interior lighting
12. (E) exterior lighting
13. (B) bomb control
14. (FL) flight controls

24 volt D.C. single wire, grd  
returns.

Power sources

1. 3-24V @ 94 A.H.

2. 4-24V @ 200 A.

3. External power.

1. 275 specific gravity at  $80^{\circ} \pm 0$ , with 4 subtracted  
at each  $10^{\circ}$  or added at each  $10^{\circ}$ .

1st battery (W.S.) will  
freeze  $10^{\circ}$  above  $0^{\circ}F$ . Full  
battery (W. 275) will freeze  
 $80^{\circ}$  below  $0^{\circ}F$ .

Disconnect grounded lead  
first after removing the  
battery fuses. (Battery

solenoid switch in nacelle  
next to battery and also  
fuse). (Letters indicate circuit,  
number is the serial number  
of wire. Letter and number  
indicate wire no.).

- Sheet 1 from Nose to 3.
- Sheet 2 from 3 to 4.
- Sheet 3 from 4 to tail.
- Sheet 4. Left wing.
- Sheet 5. Right wing.
- Sheet 6. Fabrication.

(12 diagrams in each plane).

26V for motorized instruments.

(Power circuits having fuses  
located aft of sta. 5, will

be fused in # 305, all other  
power circuits and control  
circuits will be fused in  
# 146. Except those fused  
in nacelle and shield (# 43).

(Oxygen warning, sub-cool,  
and windshield wiper motor  
fuses are in # 243) (Battery  
cut-out fuses and landing  
lights are fuse boxes in  
nacelles) (4 starters, 4 D.F.  
motors, 2 main landing  
gear motors, bomb bay  
door motor has no fuse  
for them).  $\frac{E}{I \times R} = \text{volts}$ ,  
 $I = \text{amps}$ ,  $R = \text{ohms} (\frac{E}{I \times R})$ .

flash field leads, from B  
to A lead (power to field).

Rectifier prevents the generator  
with reverse polarity from  
flowing on the line.

Generator output controlled  
by the field strength.

When adjusting voltage regulator  
never turn on generator.

26.6  $\pm$  0 Rev. Cur. R. cuts in.

10 amperes reverse current  
cuts generator off.

Reverse polarity causes voltmeter  
to read backwards,  
no reading on ammeter.

### Fuselage

Stressed skin semi-monocoque  
design with longitudinal

and bulkheads for bracing.  
74.9" long - 19.1" high  
103.9" wing span.

31,000 Empty weight.

48,226 gross weight.

1. Forward comp. - sta 1-3.

2. Pilot and necessary comp. -  
between sta. 3-4.

3. Bombay - sta. 4-5.

4. Radio comp. - sta. 5-6.

5. Waste comp. - sta. 6-7.

6. Tail Gunner's comp. sta. 7.

Bulkheads 3, 4, 5, 6 are  
solid bulkheads. Tail  
gear attached to bulkhead

7. Bulkheads 7, 8, 9, 10, 11 are

reinforced bulkheads, due  
to place of attachment  
of stabilizer and rudder.

Longitudinal stiffeners are  
made by extrusion (part  
that is pressed out and forms  
bulk angles). (longeons,  
also called off beams but  
not on ship). The skin  
and supports the forelegs.  
Always use the ship as a  
pattern for making repairs.  
Bombay truss beams are made  
of steel, bracing are made  
of aluminum alloy.

Forward  
Comp.

## Equipment

1. 2-50 flap gun mounts
- 2- upholstered chairs
- 1- C.O<sup>2</sup> hand fire extinguisher
- 1- blinker signal light
- 8- ammunition boxes
- 1- first aid kit
- 1- Navigator's table
- Navigator's astrodome
- Bomb doors control panel.

2. Lower accessory comp.  
Bomb doors and bomb  
release hinge mechanism
3. Elevated cockpit  
two bucket seats

- life preserver cushions
- 2- thermos bottles
- 1- drinking cup dispenser
- 1- Co<sup>2</sup>, 1- CCo<sup>4</sup> hand fire extinguishers.
- 1- Lewis Top gun turret
- 1- first aid kit.
- 1- hatchet

#### 4. Bombay -

- Bomb door extracting mech.
- Cook sacks and hoisting mech.
- relief tubes.
- 2 life rafts over Bombay.

5. Radio Comp. 5-6

1-50 flex gun mount  
1-ammunition box with  
flexible feed.

1-table

1-20" hand fire extinguisher

3- bucket seats with life  
preserver cushions.

2- cranks and extensions

1- first aid kit.

1- safety belt holder.

Camera pit under floor.

tool kit, mooring bits,

canvas covers for engine

motor, seat pit and life preserver

Ball turret and tail gun  
comp.

6. Waist Comp. 6-11

1- first aid kit

1- ladder

1- ball power gun turret.

2- small canvas curtains.

2- 50 flexible side gun  
mounts.

2- ammunition boxes, flexible

1- power generator

1- 50 lb fire extinguisher

Tail gun comp. 11- aft

2- 50 lb mounts.

- 2 - flex. ammunition boxes
- 1 - saddle seat adjustable.

### Emergency equipment

- 6 - hand fire extinguishers
- 3 - CO<sub>2</sub>, 2 CCL<sub>4</sub> bottles
- releasable surge panels and entrance doors for bombadier and main <sup>entrance</sup> and tail gunner.

First aid kits

bomb door and bomb release

10 - life preservers

2 - life rafts

Armor plated on bulkhead

3 - pilot, co-pilot, radio room  
and turret.

Jenkins Bearings  
Mildieum steel

Landing and tail gears.

Single oleo, cantilever type.  
(oleo - means oil and air).

oleo strut, drag strut, retracting  
strut. Turnion pin holds  
strut to structure.

.005 to .010 clearance on pins.

Release filler plug only  
one turn for letting out  
the air.

pressure in hydraulic system  
800 lbs  $\psi$ .

138 turns on retracting screw,  
screw made of steel.

retracting nut made of bronze  
(13 threads). 25 to 50 lbs pull  
to release neoprene bumpers.

To set limit switches fully retract land. Wear them back off one turn. To set lower limit turn landing gear full down then back off four turns.

(6 switches on landing gears.)

.040  $\pm$  .005 brake clearance (new)

.015 to .025 older brake clearance

$\frac{3}{16}$ " wear on brakes

Tail Gear assembly  
Single oleo, treadle bracing.  
Yoke connects piston with  
bulkhead no. 11.

$2\frac{5}{8}$ " strut showing.

420' lbs torque setting of  
anti-skidding brake.

50<sup>th</sup> tension on locking pin  
cable. (Containing switch breaks  
in circuit if wheel is not centered.)  
(5 switches on Tail gear).

$\frac{3}{8}$ " length, allowed  $\frac{1}{8}$  wear.  
(.008 clearance allowed  
between knuckle of axle  
and cylinder).

— Pitot Static System — — —

Bank and Turn — 8,000 R.P.M.

flight indicator — 15,000 R.P.M.

Magnitude of Turn — 12,000 R.P.M.

Pumps has 10 inches Hg. on  
suction line, approx case 4

checked by adjustments made  
in nacelles.

Vacuum - green & white.  
Pump and turn operates at 1.9  
inches Hg.

Auto - black.

Static - black and green.

(Autogyro Instruments).

The autogyro instrument is the  
transmission of a mechanical  
function from one place  
to another by means of  
electrical action.

Always match colors at  
connecting ends.

Inverter turns out 2 1/2 volts  
for autograph instrument, 115  
Volts also for fluorescent lamps,  
drift meter - radio compass.

De-Lee system

Pump turns turn 1 1/2  
times at normal speed.  
Turns out 9 lbs 11" pressure.  
Make sure that oil bowl  
line up properly when  
pump is changed during  
engine change.

Pump 1/2" out of round.  
(distributor valve and oil  
separator and relief valve  
in R.H. wing gap).

(Use of vacuum in camera)

operation of gyro camera  
operation of shutter, keep  
film tight, use on lens.

(25 hrs inspection of screens  
on suction Rel. valve, inspect  
restricted fittings in lines,  
flush oil separator at 150  
hour inspections.)

- Oxygen -

Always be careful  
when working on the  
oxygen system.

18 - Bottles. + 2 F-1 bottles  
later models have not  
F-1 bottles.

A9A and A12 systems.

7 Bottles in accuracy set.

8 in pilots comp.

3 in Radio comp.

1 in each turret.

Dual check valves is to prevent loss of oxygen if one bottle is not airtight.

350 P.S.I. - A9A.

095250 - 5/16" tubing.

Use modified sleeves, turn finger tight, 1/3 turn with wrench.

{ Use glydag lubricant -  
(glycerine + graphite) }

Color code - green - suit for

Enter suit - green - suit for

tubing supported and bonded every 16 inches, and with in 4 inches on each side of a fitting.

Bottles are stainless steel. Oxygen is dehydrated before it goes into the system. Never bottle out of fitting never screw fittings off of bottles.

H. P. A. indicators, bottom needle shows cylinder pressure, top needle shows oxygen flow. Service regulator needle valve with castile soap. Clean

screen and clean office  
in adapter.

A-12-Demand Regulator, in action  
you breath oxygen <sup>and air</sup> as you  
need it. on the "on" position  
you breath all oxygen.

B- Auxiliary emergency bottle  
they last 8 to 12 minutes  
(carry around bottles).

50 Hour Test - build pressure  
up to 500 P.S.I., allow to  
stand one hour. Loss can  
be allowed (1<sup>st</sup> - 1<sup>o</sup> F. or 1<sup>st</sup> -  
1<sup>o</sup> C.). Then allow to stand  
11 hours. Allow 5% loss  
for 11 hours. or 18 to 20.

Bleed pressure down to 375  
P.S.I. for the 16 hour test.  
Bleed pressure to 50 P.S.I.,  
check regulators for flow.  
Bleed pressure down and  
fill system with antifreeze

Anti-freeze system

81% ethyl alcohol  
15% glycerine } fluid  
4% methyl alcohol

20 gal cap. - supply tank.  
color code - red and white.

{ Water solution - glycol  
55% diethylene glycol  
45% ethylene glycol.

$\frac{1}{2}$ " tubing 24 S.T.  $\frac{1}{8}$  hard.

$\frac{3}{8}$ " stainless steel tubing  
around exhaust.

Radiator is  $\frac{1}{2}$ " continuous  
copper coil.

Oldag is thread lubricant  
used (oil and graphite).

Lines covered with asbestos  
painted over with neoprene  
paint.

Sliding Vane type pump  
with positive displacement.

Pump on No. 2 engine

Code color - White, black, white.

Pumps 50 to 100 P.S.F. pump  
55 to 60 gals. per hour.

pumps geared to 1 1/2 engine speed.

Be sure that cover plates are put on when pump or heaters are removed.

Low relief valve set at 300 P.S.I. valve as safety factor.

Filter located ahead exhaust (vacuum filter).

Spring loaded valves set to open at 28 to 30 P.S.I.

Turn filter after every flight while glycol is still warm (about 3 turns). Flush filter every 15 hours.

## Flushing Heating system.

Remove filter, plug inlet port and flush through outlet port with steam or hot water, blow it out with air, then flush with fluid.

Drain filter every 25 hours.

Heaters are stainless steel with high nickel content.

System operates between 200 to 300° F.

Heater put out 30,000

B.T.U. per hour.

remove heaters every 150 hours to brush outside of heaters, clean and

and inside of heater every  
75 hours.

If outside air temp. is  
 $70^{\circ}$  or more, remove the  
center heater.

Safetying done with stainless  
steel wire. Radiator in  
left hand wing gap.


Temp. of  $50^{\circ}$  or above

ignition temp. control  
in cold or off position.

Duct work covered with  $\frac{1}{4}$ "  
felt except by radiator.

# Autodyn instruments

Rotor - 

Stator - 

Phase - 

Current transmitted to motor by small springs. In the indicator the current is induced by small gold spring slip rings.

(A - ground

B - power

C - phase (yellow)

D - phase (green)

Tachometer - flyball gov.

Full pressure - diaphragm (diff. pres.)

oil pressure - Bourdon tube

Manifold press. - aneroid

Autodyn leads

app

ground

{ flap for direct mech linkage.  
at trailing edge of left wing  
"A" prefix to all wires in  
autogen system.

### Hydraulic System

Fluid is non-compressible

(Pressure in a closed area  
is transmitted equally  
and undiminished in all  
directions) - Pascal's Law.)

Pressure measure in lbs.  
per square inch.

① Supply tank must be full  
with system under pressure.

Use AN-VV-0-366A fluid  
in system

Supply tank contains baffle plate to take out swirling and air bubbles.

Never remove panel with system under pressure.

② Pump relief valves adjusted to 1150 lbs. Pump electrical driven, gear type. Pump puts out 100 gal. against 800 lbs. pressure per minute.

(Pre-lubricate a new pump under 200<sup>+</sup> pressure.)

③ Check valve follows pump in the line.

④ Oil filter - parallel type. Turn handle for flight or

Daily, drain every 100 hours  
to remove sediment.

⑤ Accumulator service valve  
used to release pressure  
in system, bypassing fluid  
back to the supply tanks.

⑥ Pressure cutout switch  
cuts in at  $575 \pm 25$  P.S.I.  
and cuts out at  $800 \pm 25$  P.S.I.

switch works on bellows  
type mechanism, which  
in turn works a lever  
that makes contact for motor  
and also warning lights.

.020 gap clearance between  
contact points.

Windshield Wiper

Yellow and white - color code  
tank on top of fuselage  
behind top turret.

A saphran Pumps, alcohol  
operated, 2 pumps. 5 gal  
tank. alcohol is fluid.

Secondary circuits —

Booster coil about 20,000  
volts to mag. for sparks  
to engine plugs.

Not direct drive starter  
energy in wheel turned

12,000 d. P.M.

Props feather at 400 P.S.I.

Holding coil also loaded for

400 P.S.I. Props unfeather

at 600 P.S.F.

Selector switches in series  
in Fuel Transfer valve.

4 main tanks and  
4 Booster pumps with  
7<sup>th</sup> pressure. Booster pumps  
on bottom of tank.

Liquidometer works as a  
autonym. measure the  
amount of gas in tanks.

— Oil system —

Most Common Grade of  
oil is 60 S.P.E. Oil pump  
supplies 70 to 80 Pounds per  
75 P.S.F. in Pressure.

PASS

P.D.I.

• • • •  
• • • • S.G.  
• • • •

FLIGHT  
INSUR.

○

ALT.

MAG. OF  
TURN

○

○

AIR.  
SPEED

FLIGHT  
IND.

○

○

BANK  
TURN

RATE OF  
CLIMB

○

NAV. PASS

○ ○  
○ ○

R.P.M.

○ ○ ○ ○  
○ ○ ○ ○  
○ ○ ○ ○  
○ ○ ○ ○

4 9

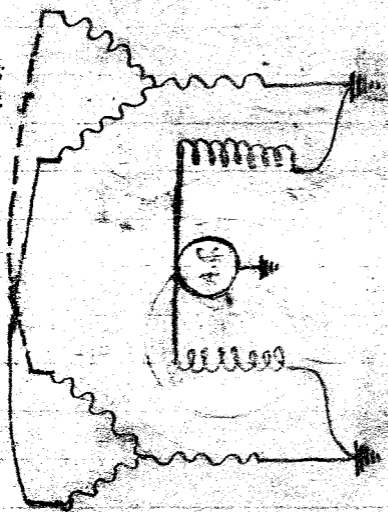
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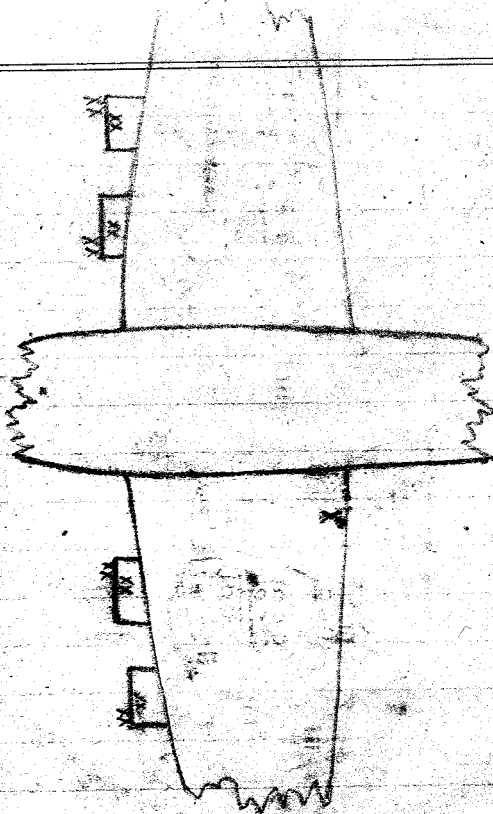
PROP. FEATHERING

Instrument Panel

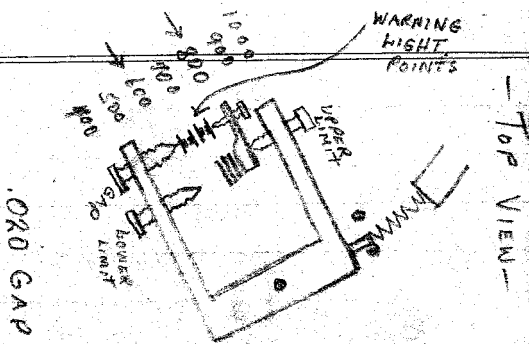
IND.

TRANS.

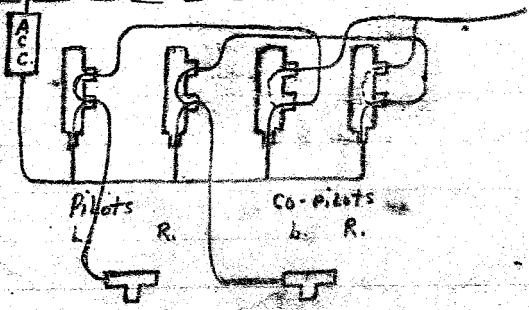


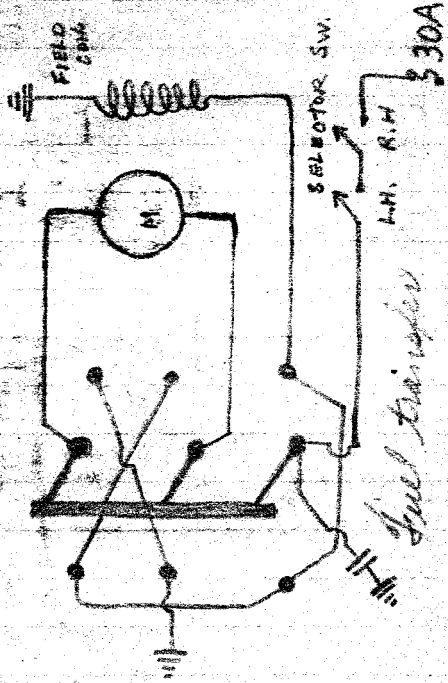


location of antenna transmitters



pressure cutout switch  
for hydraulic system

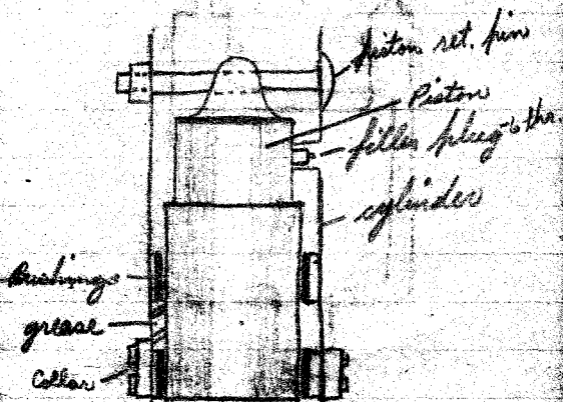






Shader # 7039 - piston / leg

.011 to .125 clearance

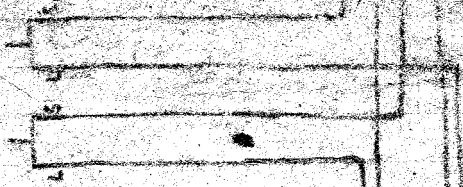
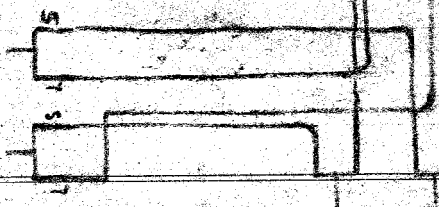


Substrate with graphite base  
grease.

oil - AN-WO-366A. Red-mineral based

1 1/2" of shut showing

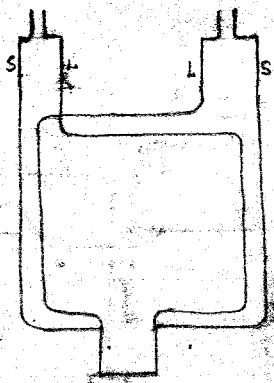
# Alison Control System



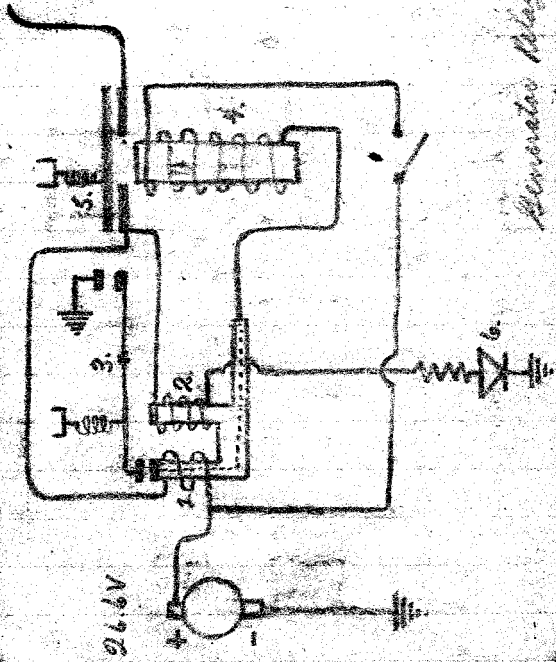
# RUDDER CONTROL SYSTEM

Pilot

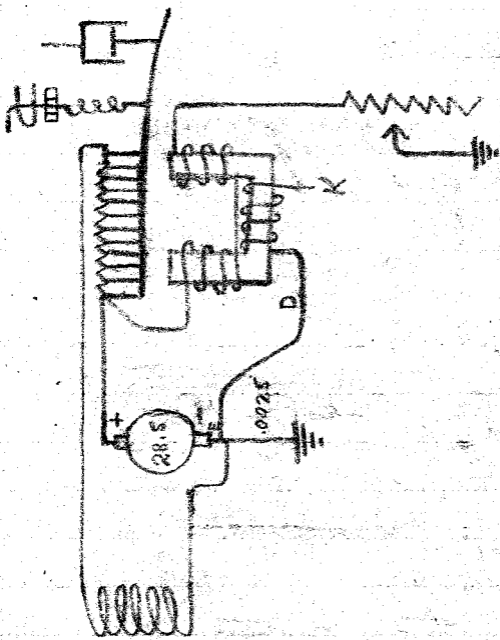
COPILOT



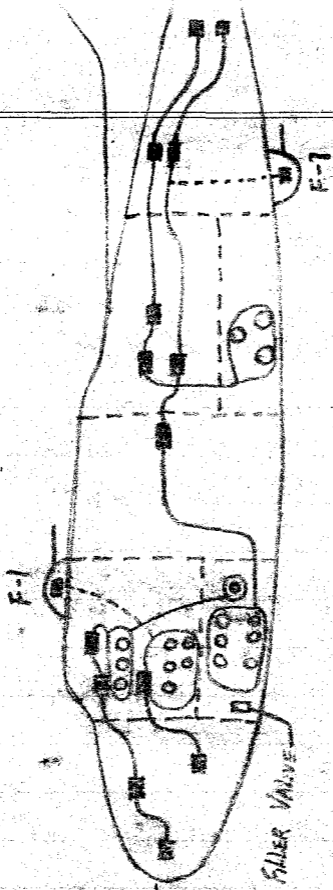
- 1- Series coil
- 2- Shunt coil
- 3- Pilot point
- 4- Main coil
- 5- Main point
- 6- C.W. Rectifier



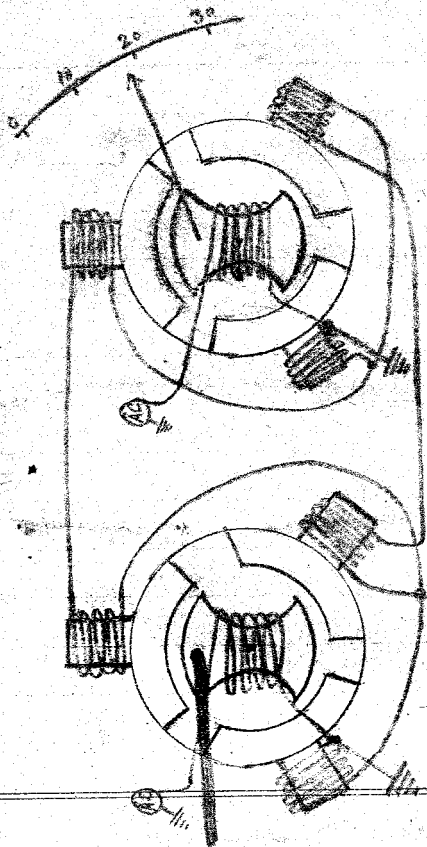
Generator Relay switch



*Generator circuit*

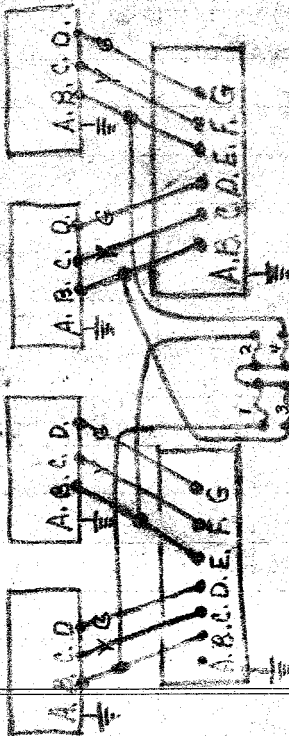


OXYGEN SYSTEM



TRANS.

IND.



DRUM IND.



SINGLE IND.



*Standard A.M. Disconnection Diagram*

Temperature regulator operates from  $60^{\circ}\text{C}$  to  $88^{\circ}\text{C}$ .

Drain 2 to 3 quarts from oil sump for a daily inspection.

$1^{\text{st}}$  check valve in front of pump keeps oil from flowing into cylinders by gravity flow, when engine is idle.

Cooler shutters are operated by oil pressure which can be thermostatically controlled. One note in adjusting makes 2 to  $3^{\circ}\text{C}$  difference in temperature.

Use benzoin for flushing  
the regulator case.

Use silver solder in  
making repairs on cooler  
jacket.

Low oil pressure may be  
due to <sup>A.</sup> low oil supply.

<sup>B.</sup> Leak in pressure line.

<sup>C.</sup> Worn or broken pumps  
or unprimed pumps. <sup>D.</sup> pressure  
relief valve improperly adj.

<sup>E.</sup> Low temperature in suction  
line, also high  
temperatures. <sup>F.</sup> Worn out  
engine with worn bearings.

<sup>G.</sup> High altitude causing  
priming. <sup>H.</sup> Air locks

Spring loaded switch on panel to pilot left is to build up pressure over 900 P.S.F. to open relief check valve and also for emergency to build up pressure.

① System relief valve picks out at 900 P.S.F. It can be connected on either side.

② Hand pump located to right of co-pilot. 2 check valves located in the pump.

③ Accumulator stores pressure, acts as shocking device. To fill accumulator use dry air built to 850 P.S.F.

then build system pressure to 800 P.S.I. Then bleed slowly.

Neoprene rings should fit loosely around piston in accumulator.

- (10) Manual shut-off valve provides cut off between main and emergency accumulators.
- (11) Selective check valve allows fluid to flow to emergency systems in normal, in servicing it allows a flow both ways. Safety in normal flight.
- (12) Emergency gage and micro switch serve emergency.

warning light on when it drops to 400 P.S.I. Don't adjust with pressure in system.

- Brake system -

adjust valves for pilot and co pilot from  $\frac{1}{8}$ " to  $1$ " from bottom of guide assembly to top of coating. To adjust emergency brake metering valve tighten the nut down until fluid starts to flow, then back off two turns.

Working brake pressure should be a min. of  $150^{\#}$  to  $170^{\#}$

Move serrated washer one notch forward.

Return boost valve reduces pressure  $\frac{1}{3}$  and expedites release of brakes.

Coil flaps - always lock coil flaps. They should travel from open to closed in 2 to 5 seconds.

adjust without pressure in system.

at 12,000<sup>th</sup> check lines for leaks, check relief valve on coil flaps and check gages.

test relief valve and  
gages at 400 P.S.I.

Test flow out out at 600 #.

check air pressure in  
both accumulators.

Four ways of bleeding  
air out of system are,

- ① bleeding at accumulator  
servicing valve.
- ② Loosen  
nut on pressure side of  
cowl flap cylinder with  
selector handle in locked pos.
- ③ bleed system through  
brake lead while hand  
pump is being worked  
and pedal compressed.

③ Work emergency metering  
valve while bleed line  
is attached.

## oil system (cont.)

Causes for high temperatures and consumption of oil.

H. Lack of oil. B. oil cooler reg. improper adjustment. C. Clogged lines, strainers, and coolers.

D. Improper engine operations.

E. Clogged vent lines.

F. Consumption due to worn engines. G. Leakage. H. Worn piston rings (improper installation).

$$C = \frac{5}{9}(F - 32)$$

$$F = \frac{9}{5}(C + 32)$$

$$F - \frac{32}{1.8} = \frac{C}{1.8}$$

} formulas for changing  $F^{\circ}$  to  $C^{\circ}$ .

Clean oil tanks at engine change.

425 } 3 main tanks.  
 313 }  
 212 }  
 850 } 270 Gals IN EACH SECTION  
 Fuel System

- 6 Main Tanks - 1900 gals.
- 2 Purifiers - 870 gals
- Tanks - 1,080 gals.

Fuel tanks are self sealing.

Use 100 octane gas or aromatic gas if the tanks have been shocked.

80 oct gas	65%	} aromatic fuel
Toluene	20%	
Xylene	15%	
Benzene	5%	

Shock tanks with zinc chromate to protect against aromatic fuel. Shock every 60 to 90 days.

Drain at Daily inspection. allowed  
10 drops.  
Check brushes at 25 hours  $\frac{1}{2}$  hour  
 $\frac{1}{2}$  hour

Boost pump is electrical  
driven. Centrifugal impeller  
type pump.

Auto out 6" to open port.  
8" to a closed port. at  
1500 R. P. M.

Plans 42-5050 and on have self  
sealing tanks with no  
metal casing.

Drain Fuel strainer daily  
and clean strainer every  
25 hours.

Fuel pump engine driven,  
sliding vane, positive  
displacement. Fuel pressure  
is 15# sq. inch.

Pump has a shear shaft.  
Remove fuel solenoid and  
drain every 100 hours.

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Retracting motors are  
series wound motors.  
They will stand a heavy  
torque and also gain  
speed.

Carburetor air filter is  
controlled by a retracting  
motor. It has one switch  
with 4 gang relays and  
a warning light.

## Superchargers

27,000 ft. is critical altitude for manifold pressure. For every 1000 ft. above 27,000 manifold pressure drops  $1\frac{1}{2}$ " hg. (B2-1 Turbo). output is 230 horsepower. takes 20 horsepower to work superchargers.

Turbine wheel turns at 21,000 R. P. M. at critical altitude.

full boost - waste gate closed  
no boost - waste gate open.

All engine control cables have a 40% tension  $\pm 10\%$

except for Governor collar,  
which is 20<sup>th</sup>.

Clearance between cooling  
cap and buckets is  $\frac{1}{8}$ "  
(.095" to .160").

Nozzle top clearance to buckets  
(.070" to .160").

Radial shake or side play  
clearance is .012"

End play is .015"

Worn bearings cause end  
play and side play.

## — Loading —

Thrust

Drag

lift.

weight.

48,726 designed weight

64,500 gross weight

(Mean aerodynamic  
chord - average line  
on which C.G. will  
fall.)

177.5" is length of m.a.c.

on which C.G. falls

(Average lift of a wing  
section - m.a.c.)

(Always use basic combat  
conditions to base

Take all guns + amm. except  
 side guns - sched. B' except  
 Prof. anti-ice fluid  
 200 gal. gas, 140 oil, 200 bag. 6-6D  
 100 lbs. - ice. comp. #10,000 ambs

loads on (d).

Nav.

Bomb.

Co-P.

Pilot.

R. operator

Eng. S.

Item	Position	Weight	C. Shift
Basic load.		37,675	31.50
Nav.	nav. station	+ 200	-.55
Tp. S.	3H	- 200	+ .40
Side S.	6D-6H	- 200	-.80
Side S.	6D-6H	- 293	- 1.20
Fluid	Radio C.	- 125	-.15
Gas	Wings	+ 17,000	—

oil savings	+1050	—
logs 6-60	+200	+50
luck 3H	+100	-20
<u>London Bomb</u>	<u>+1000</u>	<u>—</u>

+ 61,225	32.40
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- 818	2.90
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<u>60,407</u>	<u>29.50%</u>
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29.5%

19% to 32% limits

# Aluminium

24 S.R.T. Strain Hardened

No. 3 pure aluminium  
No. 2450-ST 95% al. 5% copper

S - means wrought  
aluminium. (worked out  
while still hot).

T - tempered aluminium

O - annealed "

T and O differ only in  
the cooling process.

(Copper in aluminium  
makes it unfit for  
welding and also causes  
corrosion).

Alclad and anodizing are  
two most common  
forms of protective coating.

Place rivets twice their diameter  
from edge -  $\frac{1}{4}$ " is min. distance  
from edge. Lap joint 5" diameter  
of rivets.

(Chromic acid and electrical  
current (40 volts) - solution  
used in anodizing  
aluminum oxide is  
the protective coating)

24 } stock aluminum } structural  
14 }  
A17 - rivets

non-heat treated

25	Pure aluminum
33	
45	<u>I - eye beam</u>
31	1. dinging hammers
32	2. dolly bars
51	3. rawhide mallets
	4. plastic hammer
	5. scriber
	6. easy out

$\frac{3}{32}$  - 40     $\frac{1}{8}$  - 30     $\frac{1}{32}$  - 20  
 $\frac{3}{16}$  - 10     $\frac{1}{4}$  - 5     $\frac{5}{16}$  - 0     $\frac{3}{8}$  - 4

7. snips or duck bills (straight)

8. right and left snips

9. key hole back saw

10. Tom thumb back saw

11. Files {

- half round
- mill file
- croquet file
- knife
- 3 cornered
- rattail
- mouse tail

second cut  
 single cut  
 double cut  
 smooth  
 rough

rasp file

12. Pencil divider.

13. Drills

14. Center punch

rivets sticks out  $1\frac{1}{2}$  diameters  
Bucked down to  $\frac{1}{2}$  diameters  
use die on rivet heads

Monel rivets used to rivet  
steel on firewalls, etc. (SS)

A N 4 - 16 A

A. N. - Army Navy

4 - dia. in 16th's

length 16

{ 1 - inches 1st.

{ 6 - eights 2nd

A - no cotter hole

## Handling

103' 9" wing span.

74' 9" length of fuselage

3 : 4 is ratio of L. + W.

19' 1" height

{ Prop clearance on inboard  
is 1' 5"

{ Prop clearance on outboard  
2' 5"

21' between wheels

Wing dihedral  $4\frac{1}{2}^\circ$

M.A.C. 177.5

Sweep back of wing  $8^\circ$

{ 1,277 sq. ft. area on  
top of one wing.

1-7 cone 78" from grd.  
wheel knuckle  $19\frac{1}{2}''$

3-5 cones 68" from grd  
Plane designed for  
37,000 lbs empty.

48,726 lbs loaded.

65,000 lbs. have been  
loaded on and taken  
of with it.

—Taping—

1. Always unlock the tail wheel before taping.
2. Watch out for near by planes
4. Taxi at slow speed (10 to 20 M.P.H.)
5. Importance of level ground. Use 3 men

to maneuver a plane.

b. Be careful of scratches and dents on plane.

Towing

Towing logs designed to pull 10,000<sup>##</sup>.

1. Always release parking brakes and remove the chocks first.
2. Towing line should be  $3\frac{1}{2}$  times distance of tread. (at least 75 ft. long)
3. Tow plane slowly, be careful of holes, chocks, etc.
4. With gas tanks removed

never taxi over 5 M.P.H.

5. Don't help with a tractor if using engines for taxiing.

6. Don't taxi backwards.

Two towing lugs on tail wheel, 7,000# allowable pull straight, 4000# pull at 45° and 4000# pull at 90° with plane in motion. No 90° pull if plane is standing still. Always unlock tail wheel. Towing bar shears at 7,000#. One towing

bar comes with over  
5 planes.

All engines should pull  
evenly for taxing.

Have proper tire inflation.

- Deflating tires -

E - 16 ply.

F - 14 ply.

for E 21  $\frac{1}{16}$ "

for F 20  $\frac{3}{4}$ "

Pressure is from 65 to 70<sup>psi</sup>  
in tires.

For every 1000<sup>lb</sup> weight  
increase 1<sup>lb</sup> of air in tire.

{ tail wheel inflated to  
10  $\frac{1}{2}$ " , 55<sup>psi</sup> press. average

- Mooring -

2 Wing wells on sta. 24  
good for 10,000# pull.

2 mooring knobs on knuck  
of landing gear at 10,000#  
5000# pull on tail  
wheel strut.

- Mooring kit -

{ anchor rod eye  
anchor rod  
driving rod  
mooring anchor.

Leave 16" slack in  
mooring ropes.

slack in rope } reason  
tires deflate } for 16"  
also deflate } slack

1. attach lines to mooring points.
2. Consider plane can stand 60 mile hour wind.
3. Place head to the wing.
4. Chock the wheels.
5. Lock rudder and elevator in neutral position.

## Generator Brushes

1 1/6" brushes when new,  
replace after 5/16" wear.

Generator operates 28.5 Volts.

<sup>field</sup> A-lead is return to field.

<sup>new</sup> B- goes to voltage regulator.

C. -

D. -

E. - ground

Starter brushes allowed 1/3  
wear of brushes (3/16" new).

Never use solvent unless  
at major overhaul.

• Auxiliary Generator puts  
out 28.5 and 46 amperes.

## 20 exterior lights

{ 2 landing lights  
3 fuses - 60 amp - (15 amp.) <sup>controls</sup>  
2 relays

{ 4 identification lights.  
white, red, green, amber.

1 passing light - Red (L.H. side)

6 running lights - white light  
each side of tail, 2 red  
lights on left wing tip  
2 green lights on right  
wing tip.

7 formation lights, 3 on  
dorsal and 2 on each  
side of stabilizer. (Blue)



Jacks and weight

No. 1-7 48,000 #

No. 2-6 20,000 #

No. 3-5 36,000 #

No. 4 7,000 #

No. 8 3,000 #

1. strength of structure.
2. strength of jacks.
3. Balance of plane.
4. Uniform lift while jacking.
5. Warn jacking foundation.
6. select proper jacking points
7. Importance of level plane
8. release parking brakes
9. release oleo pressure
10. tail wheel <sup>will</sup> extended when it is <sup>put</sup> under weight.

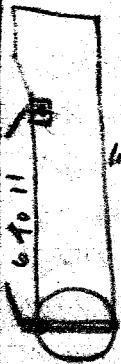
— shoring —

Use benches with sand bags on top or cradles with a thick felt.

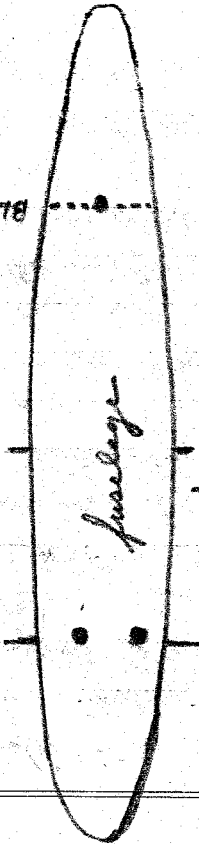
— Hoisting —

Hoist in vertical position only, except at wing connections.

1. head into wind for starting
2. Before starting engines examine record and report for C.M. and load.
3. Check oil & leakage, check fuel, check caps
4. drain fuel strainer, check



BLK 4



burner

safety wire.

5. Check bonbay tanks,  
leakage were brush over de  
sicc boots.

7. Ground plane before  
refueling.

8. Test battery.

9. Inspect tires, wheels.

10. test brake metering valve  
"5" clearing.

11. Check anti-ice fluid

12. Check props.

13. Examine windshields.

14. Examine landing gear  
tail gear.

15. Test instruments.

16. Examine red C. 0<sup>2</sup> disks.